Milestone Review Flysheet

Institution Piedmont Virginia Community College

Milestone CDR

Vehicle Properties		
Total Length (in)	108.8	
Diameter (in)	5.5	
Gross Lift Off Weigh (lb)	37.27	
Airframe Material	G12 Fiberglass	
Fin Material	G10 Fiberglass	
Coupler Length	14 in	

Stability Analysis		
Center of Pressure (in from nose)	77.7	
Center of Gravity (in from nose)	60.2	
Static Stability Margin	2.26	
Static Stability Margin (off launch rail)	2.54	
Thrust-to-Weight Ratio	7.6:1	
Rail Size and Length (in)	1515, 144	
Rail Exit Velocity	71 ft/s	

Recovery System Properties					
	Dogue Parachute				
Manufactu	ırer/Model	Sunward Gro	up Ltd / 18" Nyl	on Parachute	
Si	ze		18 in		
Altitu	de at Deployme	ent (ft)	5270		
Veloci	ty at Deploymer	nt (ft/s)	5	2	
Ter	Terminal Velocity (ft/s)			111.5	
Recovery Harness Material		aterial	Tubular Kevlar		
Harness Size/Thickness (in)		ss (in)	1/2		
Recov	ery Harness Len	gth (ft)	27		
Harness/Airframe Interfaces		1 swivel tied to each end of the harness with 2 quick links attached to each, each quick link attached to a different U-bolt on the airframe.			
Kinetic Enerfy	Section 1	Section 2	Section 3	Section 4	
of Each Section (Ft-Ibs)	875	883	1455	N/A	

Recovery Electonics		
ssile Works / RRC3 "Sport"		
Altimeter		
wo altimeters with separate eries. Use separate ejection narges for each altimeter.		
~83 hr		
7		

Motor Properties				
Motor Designation	Aerotech L1150R			
Max/Average Thrust (lb)	294.5 / 258.5			
Total Impulse (lbf-s)	801.35			
Mass Before/After Burn	8.1 / 3.9			
Liftoff Thrust (lb)	292.3			
Motor Retention	Screw-on Retainer			

Ascent Analysis		
Maximum Veloxity (ft/s)	600	
Maximum Mach Number	0.55	
Maximum Acceleration (ft/s^2)	265	
Target Apogee (From Simulations)	5282	
Stable Velocity (ft/s)	52	
Distance to Stable Velocity (ft)	7.5	·

Recovery System Properties					
	Main Parachute				
Manufactu	ırer/Model	Giant	Leap Rocketry /	Tac-1	
Si	ze		84 in		
Altitu	de at Deployme	nt (ft)	800		
Veloci	ty at Deploymer	nt (ft/s)	11	1.5	
Ter	minal Velocity (ft/s)	18.8		
Recovery Harness Material		aterial	Tubular Kevlar		
Harness Size/Thickness (in)		ss (in)	1/2		
Recov	ery Harness Len	gth (ft)	27		
Harness/Airframe Interfaces 1 swivel tied to each end of the harness quick links attached to each, each quic attached to a different U-bolt on the air			each quick link		
Kinetic Enerfy	Section 1	Section 2	Section 3	Section 4	
of Each Section (Ft-lbs)	24.9	25.1	41.4	N/A	

Recovery Electonics			
Rocket Locators (Make/Model)	Adafruit/Adafruit Ultimate GPS		
, , , , , , , , , , , , , , , , , , , ,	Breakout		
Transmitting Frequencies	902 - 928 MHz		
Black Powder Mass Drogue Chute (grams)	1.6		
Black Powder Mass Main Chute (grams)	2.8		

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	Autonomous Ground Support Equipment (MAV Teams Only)
	Overview
Capture Mechanism	
	Overview
Container Mechanism	
	Overview
Launch Rail Mechanism	***Include Description of rail locking mechanism***
	Overview
Igniter Installation Mechanism	

	Payload
	Overview
Payload 1	Roll induction and counter roll. After motor burnout, the rocket will roll twice about its long axis. While rolling, it will use a camera to detect the ground targets from the target identification challenge. After it has completed two rolls, it will roll to aim a camera at one or more of the ground targets. After that, the rocket will return to whatever roll it had after motor burnout.
	Overview
Payload 2	

	Test Plans, Status, and Results			
Ejection Charge Tests	The ejection charge test for the subscale rocket was performed successfully prior to its test flight. The ejection charge test for the full scale rocket is planned for January 27th.			
Sub-scale Test Flights	The subscale test flight was performed on December 10th; however, there was a motor anomaly which caused the rocket to crash shortly after motor ignition. The data that was gathered from the flight was analyzed and is included in the CDR report and slides. The results verified the functionality of the airframe, and provided a chance to finalize checklists for both launch and packing.			
Full-scale Test Flights	The full scale test flight is planned for February 4th at the Valley Aerospace Team launch. If that date does not work, backup launches are planned for the 5th, 11th, and 12th of February.			

Milestone Review Flysheet Institution Milestone **Additional Comments**